## Space Applications Workshop

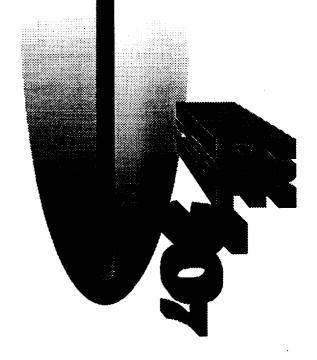
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#### AGENDA:

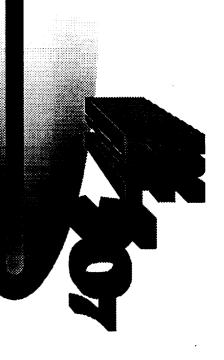
ADVOCACY FOR COTS TEST RESULTS - ELECTRICAL, C-SAM, BURN-IN MARS01 COTS SCREENING FLOW MARS01 PROGRAM/REQUIREMENTS

**VALUE ADDED ANALYSIS (Risk Reduction)** 

**VALUE ADDED ANALYSIS (Cost)** 

IMPACT of COTS++ SCREENING

SUMMARY

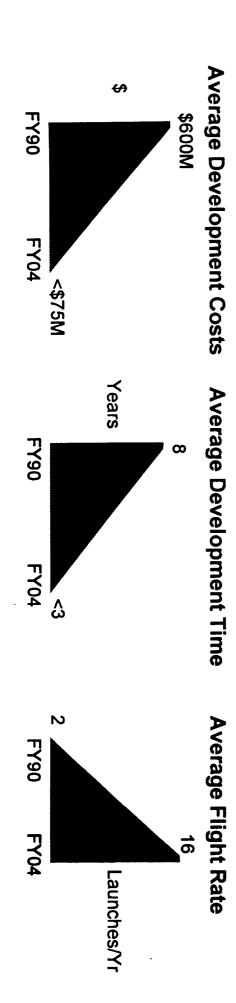


## We Have Moved From Risk Avoidance to Risk

ment:

### JPL/NASA Project Drivers:

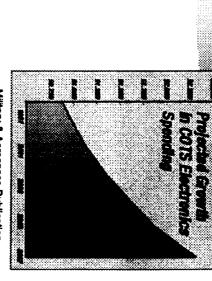
- Must infuse the latest technology (COTS is risky for High-Rel Space Application)
- Must significantly reduce development costs (COTS cost is conditional with risk)
- Must significantly reduce development time (COTS life cycle is short)





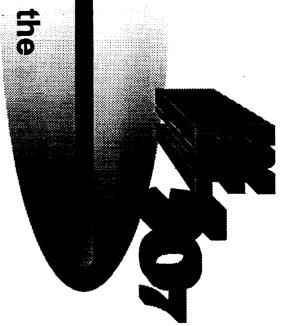
## Advocacy for Using COTS(plastic packages

- 1. State of-the-art parts are mostly available as COTS
- 2. COTS plastic parts performance (e.g. processing power & high density memories) capabilities continue to increase



Military & Aerospace Publication

- 3. COTS plastic parts enable reduction of hardware weight and volume
- 4. COTS plastic parts initial acquisition cost is less than ceramic
- 5.COTS plastic parts have been reported to demonstrate good to excellent reliability in commercial and aerospace applications
- 6. Often they are the only option when Grade 1 is not offered or available



### COTS PEM Risk Mitigation Addresses the Following Concerns:

- Narrow Temperature Range for Commercial Grade
- Plastic Assembly Quality
- Lot Non-Uniformity & Traceability
- Adequacy of Vendors Testing
- Infant Mortality
- Die Construction and Quality



## Radiation Requirements Complicates COTS for Space

Rad Hard Assurance often unknown

Energetic Particles Investigation (EPI)

Radiation requirement is unique-

Can't leverage off other high rel users like automotive

TID response depends on process-

"Positive" process changes can reduce radiation tolerance

SEE depends on circuit design and dimensions-

No good way of predicting radiation response without testing-

Commercial vendor can change these without notice

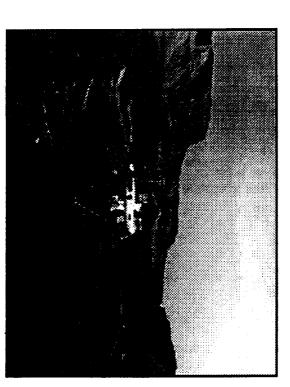
Exception is a controlled Rad Hard process line

Radiation risk mitigation techniques are often required- \$\$\$



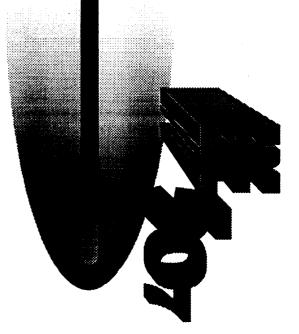
# MARS01 Pancam Plastic Parts Reliability Requirements:

- Mission Life ≤ 1 years (1500 hours operating)
- Operating Temperature (day only) = -50°C to +10°C
- Number of T/C ≈ 365
- No Assembly Board Burn-in Planned
- Outgassing is a concern
- Environmental Moisture is not critical



Lander & Rover

**JET PROPULSION LABORATORY Electronic Parts Engineering Office** (Tailored for MARS01 application/mission COTS<sup>++</sup> Plastic Infusion Baseline Flow requirements) Part Level



## DPA Results (No. of Rejects):

## Amplifier - Vendor A ADC - Vendor B

### DC-DC Converter - Vendor C

External Visual: Pass External Visual: Pass

Radiographic: Pass Radiographic: Pass

Internal Visual: Pass

SEM: Pass (1) (1/8)

SEM: Pass (0/4)

Internal Visual: Pass

**External Visual: Pass** 

Radiographic: Pass

Internal Visual: Pass

SEM: Pass (0/4)

(1) Voids found in the sidewall metalization at contact windows and was observed to be thin for one part. Although all parts were of the same date code, the dice were clearly from different processing lots

Note: Reject criteria was defined by JPL to be a potential risk to mission success.



## Initial Electrical Test Results (Pre T/C & C-SAM - No. of T

Amplifier - Vendor A ADC - Vendor B

DC-DC Converter - Vendor C

At +25°C: 0/78

At +25°C: Not tested

A+ 1700- A

At -55°C: Not tested

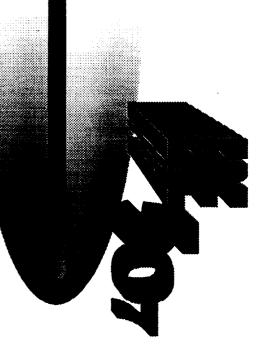
At -55°C: 0/78

At +25°C: 0/78

At -55°C: 1/78(1)

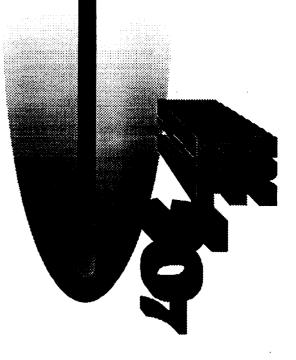
Failed parametric

Note: T/C condition = -60C to +25C (10 cycles)



# Reported Failure Mechanisms from PEM Delamination:

- Stress-induced passivation damage over the die surface
- Wire bond degradation due to shear displacement
- Accelerated metal corrosion
- Die attach adhesion
- Intermittent electricals at high temperature
- Popcorn cracking
- Die cracking



## C-SAM Results (No. of Rejects):

#### Amplifier - Vendor A

Top Side: 0/78<sup>(1)</sup>

#### ADC - Vendor B

#### .

Top Side: 30/78

#### Back Side: 8/78

Back Side: 3/78

Typical Rejects:

## DC-DC Converter - Vendor C

Top Side: 0/78

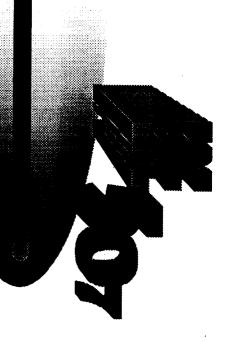
Thru Scan: 16/78





Fai

the supplier as a gel coat and is used to relieve stress of the die and improve performance. coating. These parts were not rejected. F.A. confirmed a die top coating. This was validated by mission success. (1) All units showed 100% delamination caused by a special die top Note: Units with delamination are defective and were defined by JPL to be a potential risk to



## Electrical Test Results (Pre Burn-In - No. of Re

Amplifier - Vendor A ADC - Vendor B

At +25°C: 10 (1)

At +55°C: 0

At +55°C: 0

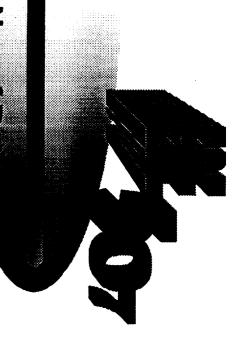
At +25°C: 0

DC-DC Converter - Vendor C

At +25°C: 2(1)

At +55°C: 1(1)

(1) Failures included parametric and functional



# Electrical Test Results (Post Burn-In - No. of Re

Amplifier - Vendor A

ADC - Vendor B

DC-DC Converter - Vendor C

At +25°C: 0

At +25°C: 0

At +25°C: 0

At -55°C: 0

At -55°C: 3(1)

At -55°C: 0

(1) Failures were parametric and functional

was calculated to simulate 1500 hrs at -10C using a T acceleration factor of 21 & Note: Burn-In Conditions = Dynamic at 72 hrs, @+55C, @max rated Vdd. This condition Ea=.33ev. The 3 burn-in circuits simulated the actual operation of the parts.



## Electrical Test Results (QCI - No. of Rejects)

## Amplifier - Vendor A ADC - Vendor B

DC-DC Converter - Vendor C

At +25°C: 0

At -55°C: 0

At +25°C: 0

At -55°C: 0

At +25°C: 0

At -55°C: 0

Note: All parts passed (ss = 10 good parts/part type)

parts. condition was calculated to simulate additional 1500 hrs at -10C using a T acceleration Note: Burn-In Conditions = Extended dynamic at 72 hrs, @+55C, @max rated Vdd. This factor of 21 & Ea=.33ev. The 3 burn-in circuits simulated the actual operation of the

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## Circuit Card Assembly (CCA) Risk Reduction

ADC - Vendor B

Unit yield: 75/78

Amplifier - Vendor A

Unit yield: 61/78

DC-DC Converter - Vendor C

Unit yield: 31/78

W.C.Failure Rate Expected Before Screen (COTS):

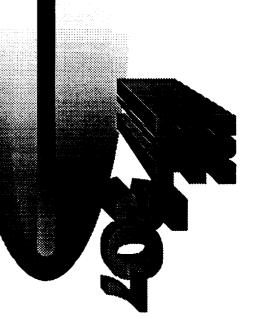
=  $100\%(1-[75/78^{1} \times 31/78^{1} \times 61/78^{1} \times 100/100^{1} \times .....])$  or  $\leq 70\%$ 

W.C.Failure Rate Expected After JPL Screen ( COTS++):

=  $100\%(1-[.990^1 \times .985^1 \times .950^1 \times .100/100^1....])$  or  $\leq 8\%$ 

Potential Risk of failure has been reduced by ≈ 62%

rejects as well as the number of package related defects. Rejects and defects were rated as equal Note: Vendor B product is potentially more at risk because of high number of pre and post BI



## **VALUE ADDED ANALYSIS (Cost):**

Amplifier - Vendor A	
ADC - Vendor B	
DC-DC Converter - Vendor C	

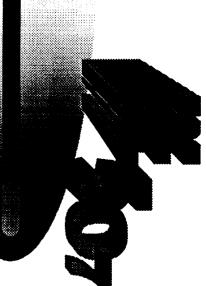
Cost:	\$.260k	\$1.8k	\$.350k
Part Screening Cost:	\$6.8k	\$13.8k	\$6.3K
Engineering O/H Cost:	\$2.0k	\$2.5k	\$2.0H
Value added for screening/CCA:	\$8.8k/9 +	\$8.8k/9 + \$16.3k/9 +\$8.3k/9	= \$3.7k

**Before Screen: Risk of Failure Cost** 

\$30k(all material & labor) x 9 x .70 f.r. = \$189K

with Screen:

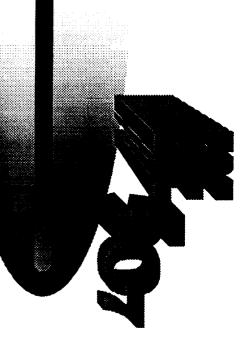
Risk of Failure Cost (\$30k + \$33.4k) x 9 x .08 f.r. = \$45.6k (>400% Potential Savings)



# COTS<sup>++</sup> PEM Upscreen Impact on Risk Mitigation

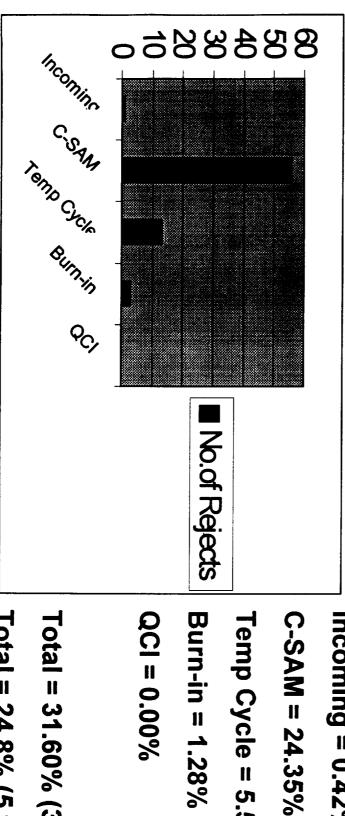
	Amplifier	ADC	DC-DC Converter
<ul> <li>Narrow Temperature Range for Commercial Grade</li> </ul>	-		<b>ຜ</b>
Plastic Assembly Quality	ω	ဖ	9
<ul> <li>Lot Non- Uniformity &amp; Traceability</li> </ul>	>	9	ယ
Adequacy of Vendors Testing	- <b>-</b>	9	ယ
Infant Mortality	_	ဖ	-
<ul> <li>Die Construction and Quality</li> </ul>	<b>-</b>	->	-
Total Score COTS⁺⁺ Impact on Lowering Risk	Low	38 High	20 High

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## Summary/Conclusions:

## COTS<sup>++</sup> Upscreening Results



Incoming = 0.42%

**Temp Cycle = 5.55%** 

**Burn-in = 1.28%** 

QCI = 0.00%

Total = 24.8% (5 types)Total = 31.60% (3 types)

## JET PR

	JET PROPULSION LABORATORY Electronic Parts Engineering Office	ON LABORAT s Engineering	ORY		
COTS	COTS <sup>++</sup> Upscreening Rejects by Part & Vendo	Rejects by	Part & Vendor		
	Amplifier-Vend. A	ADC-Vend. B	DC-DC ConVend.C	Voltage C-Vend.A	S.Regulator-Vend.B*
DPA:	0/4	1/8	0/4	In progress	In Progress
ncoming:	0/78	n/a	1/78	0/80	8/80
C-SAM:	3/78	38/78	16/77	4/80	0/80
mp Cycle:	0/78	10/78	3/77	0/80	3/72
Bum-In:	0/78	3/68	0/74	0/80	9/69
QC:	0/10	0/10	0/10	0/10	0/10
Total:	3/78	51/78	20/78	4/80	20/80

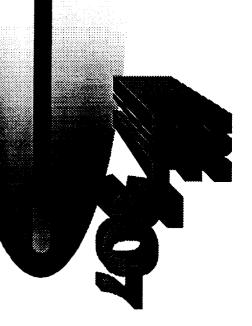
Temp Cycle:

Bum-in:

C-SAM:

Incoming:

<sup>\*</sup> New data with different part numbers



## Summary/Conclusions:

- been validated from the results of the tailored screening flow used The concerns/risks anticipated with using COTS PEMS have
- potential risk of failure for the MARs01 CCA by approximately The tailored screening flow used has significantly reduced the
- launcn). screened parts has been reduced by a much as 400% (before The cost of failure for future CCAs manufactured with the
- screening/characterization will jeopardize any Project until the unknown risks/concerns are understood and mitigated Using COTS PEMs without any value added